Docket No.: 2901683.19

1. (currently amended) A method for manufacturing aluminium aluminum alloy parts with precipitation hardening comprising:

AMENDMENTS TO THE CLAIMS

subjecting at least two elements made from the same alloy or different alloys to heat treatment at a temperature T for at least  $2t_1$ , wherein  $t_1$  comprises a minimum treatment duration at temperature T leading to a specific melting peak energy defined by <u>Differential Scanning</u> Calorimetry AED and less than 1 J/g;

friction stir welding said at least two elements, and thereafter; conducting a solution heat treatment, and quenching welded parts.

- 2. (original) A method according to claim 1, wherein the specific melting peak energy is less than 0.5 J/g.
- 3. (original) A method according to claim 2, wherein the specific melting peak energy is less than 0.1 J/g.
- 4. (original) A method according to claim 1, wherein the temperature T is less than the alloy burning temperature by not more than 20°C, or if different alloys are used, the lowest burning temperature of these alloys.
- 5. (original) A method according to claim 1, wherein the burning temperature of the alloy is less than 500°C, and the treatment duration is at least 24 h.
- 6. (original) A method according to claim 5, wherein the treatment duration is at least 48 h.
- 7. (original) A method according to claim 1, wherein the heat treatment is done at a homogenization stage before rolling, extrusion, or forging.

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8. (original) A method according to claim 1, wherein the heat treatment is reheating

between two hot rolling, extrusion, or forging passes.

(original) A method according to claim 1, wherein the heat treatment is conducted on a 9.

partly finished rolled or forged product before welding.

(original) A method according to claim 9, wherein the heat treatment is followed by 10.

quenching.

(original) A method according to claim 1, wherein at least one of the alloys is a 2024 11.

allow having a manganese content by weight of less than about 0.3%.

(original) A method according to claim 1, wherein at least one alloys comprises a 12.

copper-containing alloy of the 7xxx series having a chromium content by weight of less than

about 0.15%, and a zirconium content by weight of less than about 0.09%.

(original) A method according to claim 12, wherein the copper content is at least about 13.

0.5%.

(original) A method according to claim 1, wherein inert gas is flushed over the surface of 14.

a welding zone, during welding.

(withdrawn and currently amended) A part comprising at least two elements made from 15.

aluminium aluminum alloy with precipitation hardening, welded by friction stir welding and

treated after welding by solution heat treatment and quenching, wherein the grain size in a

welded zone of said part is less than about 200 µm after solution heat treatment and quenching,

and wherein at least one of said elements is made from a copper-containing alloy of the 7xxx

series with a chromium content of less that 0.15 wt-% and a zirconium content of less than 0.09

wt-%.

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16. (withdrawn) An aeronautical construction comprising a part as claimed in claim 15.

17. (withdrawn) A friction stir welded part, wherein in a welded zone thereof, the micrography comprises a fine crystalline structure with a relatively homogenous grain size

between 50 and 200 µm.

18. (withdrawn) A friction stir welded part, of claim 17, having an average grain size in said

welded zone on the order of 120  $\mu m$ .

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